



SUPPLEMENTAL TYPE CERTIFICATE

10043724

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AIRCRAFT STRUCTURES INTERNATIONAL Corp.

**ENID WOODRING REGIONAL AIRPORT, HANGAR 31
RR 5, BOX 41B
ENID OK 73701
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number : EASA.IM.A.226
Type Certificate Holder : CESSNA AIRCRAFT COMPANY
Type Design - Model : CESSNA 208/208B
Original STC Number : FAA STC SA11115SC**

Description of Design Change:

Installation of a Standby Flap Travel Limiting System in accordance with FAA STC SA11115SC.

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements / the following paragraph(s) at a later amendment: FAR 23.305(a)(b), Amdt. 45, FAR 23.1309(d)(e)(f), Amdt. 49.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 19 February 2013

European Aviation Safety Agency
Paul HATTON
Project Certification Manager

Note:
The following numbers are listed on the certificate:
EASA current Project Number: 0010022179-001

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Associated Technical Documentation:

Master Data List ASIC208-6000-01MDL, Revision D, released September 22, 2011;
Instructions for Continued Airworthiness, ASIC-002ICA, Revision C, released January 6, 2012;
AFM Supplement ASIC208-6000-29, Revision IR, dated December 13, 2012;
or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

The installer must determine whether this design change is compatible with previously approved modifications.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

This Supplemental Type Certificate is approved only for the product configuration as defined in the approved design data referred to in the paragraph #Description".

The installation of this modification by third persons is subject to written permission of the approval holder.

- end -

Note:
The following numbers are listed on the certificate:
EASA current Project Number: 0010022179-001

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