

**Aircraft Structures International Corp.  
Enid Woodring Regional Airport  
Enid, OK 73701**

**INSTRUCTIONS FOR CONTINUED AIRWORTHINESS**

For  
Cargo Pod Lighting Installation for Cessna 208 and 208B

**Document No.: 02010000-ICA**

**Revision A**

**Date: May 8, 2012**

**Applicable to:**

**Cessna 208 and 208B**

**ODA Project Number: ST1819DRB-A**

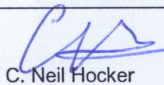
The information in the Instruction for Continued Airworthiness is FAA accepted material and complies with 14 CFR 23.1529, Instructions for Continued Airworthiness. It supersedes or adds to that provided in the Maintenance Manual for the Cessna 208 and 208B Aircraft, only where covered in the items contained herein. For limitations and procedures not contained in the Supplement, consult the Component Maintenance Manual, or other approved aircraft data.

Document Title: **Instructions for Continued Airworthiness**

Prepared By: Brian C. Adamson

Reviewed By: C. Neil Hocker

Updates to the ICA will be made by Aircraft Structures International Corp. Updates will be listed in the log of revisions and the effective pages will be listed below.

Log of Revisions				
REV. NO.	EFFECTED PAGE(S)	DESCRIPTION	DATE	APPROVED BY
Orig. Issue	All	Initial Release	02/02/2012	Martin W. McCaslin
A	All	Incorporated AEG comments	05/08/2012	 C. Neil Hocker

**LIST OF EFFECTIVE PAGES**

Page	Date	Rev
1	05/08/2012	A
2	05/08/2012	A
3	05/08/2012	A
4	05/08/2012	A
5	05/08/2012	A
6	05/08/2012	A
7	05/08/2012	A
8	05/08/2012	A
9	05/08/2012	A
10	05/08/2012	A
11	05/08/2012	A
12	05/08/2012	A
13	05/08/2012	A
14	05/08/2012	A
15	05/08/2012	A
16	05/08/2012	A
17	05/08/2012	A

## TABLE OF CONTENTS

DESCRIPTION	PAGE
REVISION PAGE .....	2
TABLE OF CONTENTS .....	3
ABBREVIATIONS AND DEFINITIONS.....	4
I     MANUAL REVISION AND DISTRIBUTION .....	5
1.0   INTRODUCTION .....	5
2.0   INSPECTION REQUIREMENTS AND OVERHAUL SCHEDULE .....	7
3.0   DIMENSION AND ACCESS:.....	7
4.0   LIFTING AND SHORING.....	7
5.0   LEVELING AND WEIGHING .....	7
6.0   TOWING AND TAXIING .....	8
7.0   PARKING AND MOORING .....	8
8.0   PLACARDS AND MARKINGS .....	8
9.0   MAINTENANCE INFORMATION .....	8
10.0  AIRWORTHINESS LIMITATIONS.....	9
APPENDIX A.....	10

## ABBREVIATIONS AND DEFINITIONS

Abbreviations	Definitions
AML	FAA Approved Model List (AML)
Detailed Inspection (DET)	An intensive examination of a specific item, installation or assembly to detect damage, failure or irregularity. Available lighting is normally supplemented with a direct source of good lighting at an intensity deemed appropriate. Inspection aids such as mirrors, magnifying lenses, etc. may be necessary. Surface cleaning and elaborate access procedures may be required.
FAA	Federal Aviation Administration
FAA MIDO	FAA Manufacturing Inspection District Office
General Visual Inspection (GVI)	A visual examination of an interior or exterior area, installation or assembly to detect obvious damage, failure or irregularity. This level of inspection is made from within touching distance unless otherwise specified. A mirror may be necessary to enhance visual access to all exposed surfaces in the inspection area. This level of inspection is made under normally available lighting conditions such as daylight, hanger lighting, flashlight or droplight and may require removal or opening of access panels or doors. Stands, ladders or platforms may be required to gain proximity to the area being checked.
ICA	Instructions for Continued Airworthiness
PMA	Parts Manufacturer Approval
STC	Supplement Type Certificate

## I MANUAL REVISION AND DISTRIBUTION

When this document requires revision it will be reissued in whole or in part. If issued in part, the revision level will be noted on the pages affected and on the List of Effective Pages. When the document requires substantial revision, it will be re-issued in whole. Aircraft Structures International will provide copies of this document to all registered operators of the equipment, and will provide revision service to all registered users of the equipment. Where reference is made to a Cessna 208 or 208B Manual it is the Manual current at the date of issue of these Instructions for Continued Airworthiness.

To register as a user or to request copies of this document contact:  
Aircraft Structures International Corp.  
1026 S 66th Street  
Enid, Oklahoma  
73701-9660 USA

Phone: (580) 242-5907  
Fax: (580) 242-5834  
E-mail: Ralph Mitten: [rmitten@asic.aero](mailto:rmitten@asic.aero)

### 1.0 INTRODUCTION

The purpose of these Instructions for Continued Airworthiness (ICA) is to provide the maintenance technician with the information necessary to ensure the continued airworthiness of the installation of an internal cargo pod lighting system per Aircraft Structures International Corp Installation Data List DL1819M02 Rev A or later FAA approved revision, in Cessna 208 and 208B.

Modifications to an aircraft obligates the operator to include the maintenance information provided by this document into the operator's aircraft Maintenance Manual and operator's aircraft scheduled maintenance program. This document defines supplementary maintenance operations and frequencies recommended by Aircraft Structures International Corp., to ensure the aircraft's airworthiness.

The information contained herein addresses the requirements specified in 14 CFR 23.1529, Instructions for Continues Airworthiness and supplements the basic aircraft maintenance manual only in those areas listed as pertains to the an internal cargo pod lighting system per the Aircraft Structures International Corp Installation Data List DL1819M02. For limitations and procedures not contained in this supplement, consult the basic aircraft maintenance manual.

#### Data

All information to support the continued airworthiness of this modification is contained in:

Aircraft Structures International Corp. Installation Data List DL1819M02 Rev A or later FAA approved revision.

This project will install an internal cargo pod lighting system in the Cessna 208 and 208B aircraft. The 208 and 208B have a three and four bay cargo pods, respectively. A single light strip will be installed in each cargo pod bay and will be illuminated whenever the cabin cargo lights are lit. In addition, a timer will be installed and programmed to shut the LED light strip off after 30 minutes of operation.

### **Design Change Control**

All data and changes to the parts and assemblies will be tracked per Installation Data List DL1819M02 Rev A or later FAA approved revision.

### **Applicable Aircraft**

Cessna 208 and 208B



**2.0 INSPECTION REQUIREMENTS AND OVERHAUL SCHEDULE**

This alteration does not change the aircraft manufacturer’s maintenance and inspection schedule.

In the event a system component failure occurs or the component does not perform its intended function, the component should be removed and replaced. Contact Aircraft Structures International Corp. for information regarding FAA approved replacement parts.

Perform the following general visual inspections annually.

**Table 2.0A**

<b>Task Code</b>	<b>Task</b>	<b>Interval</b>	<b>Mech</b>	<b>Insp</b>	<b>Date</b>
2.0.1	Perform a general visual inspection for corrosion, wear and tear, attachment condition, and loose items.	12 Months			

The following general items should be performed after removal / reinstallation of a system component and as a periodic maintenance inspection each time the associated area is inspected as a part of the normal aircraft inspection plan.

**Table 2.0B**

<b>Task Code</b>	<b>Task</b>	<b>Interval</b>	<b>Mech</b>	<b>Insp</b>	<b>Date</b>
2.0.2	Verify that all components are properly secured in their respective locations.	When cargo pod’s interiors are inspected per operator’s Inspection Plan			
2.0.3	Verify that connecting cables and associated wiring is free of damage or wear.				
2.0.4	Perform a functional test of all lights.				

**3.0 DIMENSION AND ACCESS:**

The installation of the cargo pod lighting system does not change the dimensions of the aircraft or alter the access to any existing aircraft system.

**4.0 LIFTING AND SHORING**

No change.

**5.0 LEVELING AND WEIGHING**

No change.

## 6.0 TOWING AND TAXIING

No change.

## 7.0 PARKING AND MOORING

No change.

## 8.0 PLACARDS AND MARKINGS

N/A

## 9.0 MAINTENANCE INFORMATION

With the exception to the wiring, all other parts not specifically covered by the aircraft maintenance program, are to be maintained "On Condition". Contact Aircraft Structural International for replacement parts, and other warranty service. The LED cargo pod lights and timer are not required for continued safe flight and landing.

When the cabin cargo lights are on, the cargo bay LED lights will be lit for 30 minutes. Timers in the circuit are programmed to turn the cargo bay LED light strips off after 30 minutes of operation.

**For failure of the lights to illuminate** check:

- 1) The circuit breaker has not interrupted the circuit.
- 2) For loose electrical connections.
- 3) For broken or chaffed wiring.
- 4) Broken or damaged light assemblies.
- 5) Check for damage electrical timers and switches.

For disassembly and re-installation of the LED Cargo Pod Lighting required for steps 3, 4 and 5, refer to Appendix A, Figures A-1 thru A-4 for the Cessna 208 and Figures A-5 thru A-8 for the Cessna 208B.

**For failure of lights to turn off:**

- 1) Replace timers

For disassembly and re-installation of the LED Cargo Pod Lighting required refer to Appendix A, Figures A-1 thru A-4 for the Cessna 208 and Figures A-5 thru A-8 for the Cessna 208B.



## 10.0 AIRWORTHINESS LIMITATIONS

The information contained herein supplements the basic Maintenance Manuals only in those areas listed, when the aircraft is modified in accordance with Aircraft Structures International Corp. Installation Data List DL1819M02 Rev. (IR) or later FAA approved revision. For limitations and procedures not contained in this supplement, consult the basic Maintenance Manuals.

The Airworthiness Limitations section is FAA approved and specifies maintenance required under Sec. 43.16 and 91.403 of the Federal Aviation Regulations unless an alternative program has been FAA approved.

AIRWORTHINESS LIMITATIONS - LOG OF REVISIONS			
REV.	EFFECTED PAGE (s)	DESCRIPTION of REVISION	DATE
(IR)	All	Initial Release	02/02/2012

### AIRWORTHINESS LIMITATIONS

There are no Airworthiness Limitations associated with this installation.

GENERAL NOTES:		REVISION HISTORY			
ZONE	REV	DESCRIPTION	DATE	APPROVED	
<p><b>1. READ ALL INSTRUCTIONS PRIOR TO BEGINNING MODIFICATION</b></p> <p>2. THIS DRAWING WAS PRODUCED BY COMPUTER GRAPHICS AND IS NOT TO BE MANUALLY CHANGED.</p> <p>3. ALL WORKMANSHIP AND INSTALLATION PROCEDURES ARE TO BE PERFORMED IN ACCORDANCE WITH THE STRUCTURAL REPAIR AND MAINTENANCE MANUALS.</p>					
<p>4. ALL FABRICATED ALUMINUM PARTS ARE TO BE CHEMICAL CONVERSION COATED IN ACCORDANCE WITH MIL-DTL-5541.</p> <p>5. ALL FABRICATED ALUMINUM PARTS ARE TO BE EPOXY PRIMED IN ACCORDANCE WITH MIL-PRF-23377 EXCEPT AS NOTED.</p> <p>6. BREAK SHARP EDGES AND ON ALL ALUMINUM FABRICATED PARTS 0.003-0.006 INCHES.</p> <p>7. ADJUST FASTENER LENGTH AS REQUIRED WHERE LENGTH IS NOT SPECIFIED.</p> <p>8. SOLID RIVETS ARE TO BE INSTALLED PER NAS47196 AND BLIND RIVETS ARE TO BE INSTALLED PER NAS47195.</p> <p>9. ARTICLES MANUFACTURED BY ASIC WILL BE PERMANENTLY AND LEGIBLY IDENTIFIED IN ACCORDANCE WITH 14 CFR PART 45.15(a), MARCH 1, 2010 USING INDELIBLE INK.</p> <p>10. MAINTAIN MINIMUM 2D EDGE DISTANCE AND MINIMUM 4D CENTER TO CENTER FOR ALL FASTENERS.</p> <p>11. DIMENSIONS AND PART NUMBERS IN PARENTHESIS ( ) ARE FOR REFERENCE ONLY.</p> <p>12. DISCONNECT BATTERY.</p> <p>13. REFER TO DWG 02333012, ACCOMPLISH NOTES 11 THROUGH 15.</p> <p>14. REFER TO DWG NO. 02333020, ACCOMPLISH NOTES 12 AND 13. REFERENCE NOTE 14 FOR MOUNTING LOCATIONS OF ALL PARTS.</p> <p>15. INSTALL FWD MOUNT P/N 02333030-3 IN FWD CARGO POD BAY, AND FWD MOUNTS P/N 02333030-1 IN CTR AND AFT CARGO POD BAYS. SEE FIG. 1 AND DETAILS A, B, AND C. DO NOT TIGHTEN NUTS. NOTE: SOME CARGO PODS HAVE A LIP THAT HANGS OVER THE DOOR FRAME. LONGER SCREWS THROUGH THE CARGO POD DOOR FRAME AND ADDITIONAL WASHERS BETWEEN THE FWD MOUNT AND THE DOOR FRAME MAY BE NEEDED.</p> <p>16. IN THE FORWARD CARGO POD BAY, INSTALL LIGHT P/N 6900311-000 (L ASI 3) WITH WIRE HARNESS ASSY. P/N 02333032-7 (J ASI 7) AND TIGHTEN NUTS. SEE FIG. 2, 3, DETAIL A, AND REF. DWG 02333024, FIG. 1.</p> <p>17. TEMPORARILY INSTALL LIGHT P/N 6900311-000 (L ASI 2), FWD MOUNT P/N 02333030-1 AND AFT MOUNT P/N 02333030-2 INTO THE CTR CARGO POD BAY. SEE DETAILS B AND E.</p> <p>18. BEND TUBE P/N 02333041-7 AT 16" FROM THE AFT END OF THE TUBE TO MATCH THE INSIDE CONTOUR OF THE CARGO POD AS SHOWN IN FIG. 4. MINIMUM BEND RADIUS IAW AC43.13 1-8, TABLE 9.2.</p> <p>19. INSTALL TUBE P/N 02333041-7 INTO THE HOLE IN FWD MOUNT P/N 02333030-1 IN THE CTR CARGO POD BAY. SEE FIG. 5.</p> <p>20. WHERE THE TUBE P/N 02333041-7 OVERLAPS THE AFT MOUNT P/N 02333030-2 IN THE FWD CARGO POD BAY, MARK TUBE P/N 02333041-7 AT THE AFT EDGE OF THAT MOUNT. SEE FIG. 6. REMOVE TUBE P/N 02333041-7, CUT IT 3/16" LONGER THAN THE MARK AND DEBURR.</p> <p>21. REMOVE THE LIGHT P/N 6900311-000 (L ASI 2) IN THE CTR CARGO POD BAY TO HELP INSTALL TUBE P/N 02333041-7. INSERT WIRE HARNESS ASSY 02333032-7 (J ASI 7) INTO TUBE P/N 02333041-7, THEN INSTALL TUBE WITH MOUNTING HARDWARE AND THE FWD MOUNT IN THE CTR CARGO POD BAY. SEE FIG. 3, REF. DETAILS A, B, D, AND DWG 02333024 FIG. 1.</p> <p>22. CUT WIRE HARNESS P/N 02333032-7 (J ASI 7) ONE INCH LONGER THAN THE FWD MOUNT IN THE CTR CARGO POD. STRIP END OF WIRE APPROX. .200" AND INSTALL CONNECTOR P/N 6100295-000 (P ASI 3). SEE FIG. 7 AND REF. DWG 02333024, FIG. 1.</p> <p>23. INSTALL LIGHT P/N 6900311-000 (L ASI 2) WITH WIRE HARNESS ASSY. P/N 02333032-5 (J ASI 5) AND TIGHTEN NUTS. SEE FIG. 5, 7, DETAILS B, E AND REF. DWG. 02333024, FIG. 1.</p> <p>24. TEMPORARILY ASSEMBLE THE TIMER P/N 6900329-000 (T ASI 1), LIGHT P/N 6900311-000 (L ASI 1) AND MOUNTS P/N'S 02333030-1, -2 AND -4. INSTALL THEM INTO THE AFT CARGO POD. SEE DETAILS C AND F. DO NOT TIGHTEN NUTS.</p> <p>25. DRILL Ø.193 HOLE IN THE CTR CARGO POD BULKHEAD. INSTALL BRACKET P/N MS9600-54. SEE FIG. 8 AND REF. DETAIL C. NOTE: INSTALL BOLT P/N AN3-4 TO HOLD THE ADEL CLAMP AT THIS TIME.</p> <p>26. INSTALL TUBE P/N 02333041-5 INTO THE HOLE IN FWD MOUNT P/N 02333030-1 IN THE AFT CARGO POD BAY. SEE FIG. 5. ROUTE TUBE TO THE AFT MOUNT IN THE CTR CARGO POD BAY. REF. DETAIL G. NOTE: SOME HAND FORMING WILL BE NEEDED. MINIMUM BEND RADIUS IAW AC43.13 1-8, TABLE 9.2.</p> <p>27. WHERE THE TUBE P/N 02333041-5 OVERLAPS THE AFT MOUNT P/N 02333030-2 IN THE CTR CARGO POD BAY, MARK TUBE P/N 02333041-5 AT THE AFT EDGE OF THAT MOUNT. SEE FIG. 6. REMOVE TUBE P/N 02333041-5, CUT IT 1/4" LONGER THAN THE MARK AND DE-BURR.</p> <p>28. DRILL A Ø.437" HOLE FOR AN837-D4 FITTING IN BOTTOM SKIN OF THE AIRCRAFT (AFT CARGO POD BAY). SEE FIG. 9 AND REF. DETAIL H FOR LOCATION.</p> <p>29. INSTALL AN837-D4 FITTING AND EXIT TUBE ASSY P/N 02333031-1. REF. DETAIL H. NOTE: LEAVE THE FITTING LOOSE TO HELP INSTALL TUBE ASSY. P/N 02333031-3.</p> <p>30. TEMPORARILY INSTALL TUBE ASSY P/N 02333031-3 ONTO THE AN FITTING. WHERE THE TUBE P/N 02333031-3 OVERLAPS THE AFT MOUNT P/N 02333030-2 IN THE AFT CARGO POD BAY, MARK TUBE P/N 02333031-3 AT THE AFT EDGE OF THAT MOUNT. SEE FIG 6 AND REF. DETAIL H. REMOVE TUBE P/N 02333031-3, CUT IT 3/16" LONGER THAN THE MARK AND DE-BURR.</p> <p>31. REMOVE THE TIMER P/N 6900329-000 (T ASI 1) AND LIGHT P/N 6900311-000 (L ASI 1). INSERT WIRE HARNESS ASSY 02333032-5 (J ASI 5) INTO TUBE P/N 02333041-5 AND INSTALL TUBE WITH MOUNTING HARDWARE AND THE FWD LIGHT MOUNT IN THE AFT CARGO POD BAY. SEE FIG. 8 AND 10, REF. DETAILS C, G, AND I.</p> <p>32. CUT WIRE HARNESS P/N 02333032-5 (J ASI 5) 1" LONGER THAN THE FWD MOUNT IN THE AFT CARGO POD. STRIP END OF WIRE APPROX. .200" AND INSTALL CONNECTOR P/N 6100295-000 (P ASI 2). SEE FIG. 10 AND REF DWG. 02333024-1.</p> <p>33. IN THE AFT CARGO POD BAY INSERT WIRE HARNESS P/N 02333032-3 (J ASI 3) INTO THE AFT MOUNT THEN THROUGH TUBE ASSY P/N 02333031-3, AN837-D4 FITTING, AND EXIT TUBE ASSY P/N 02333031-1. CONNECT THE TIMER P/N 6900329-000 (T ASI 1) TO WIRE HARNESS P/N 02333032-3 (J ASI 3) AND INSTALL MOUNT P/N 02333030-4 ON THE TIMER. CONNECT THE LIGHT P/N 6900311-000 (L ASI 1) TO THE TIMER P/N 6900329-000 (T ASI 1). CONNECT THE CONNECTOR P/N 6100295-000 (P ASI 2) TO THE LIGHT P/N 6900311-000 (L ASI 1). INSTALL THESE PARTS AND THEIR MOUNTING HARDWARE IN THE AFT CARGO POD BAY AT THE SAME TIME. TIGHTEN NUTS. SEE FIG. 10, DETAILS C, F, H, AND REF. DWG 02333024-1.</p> <p>34. INSTALL 3 WIRE MOUNTS P/N S2034-1 IN BULKHEADS AND LONGERON. SEE FIG. 11 AND REF DWG 02333022.</p> <p>35. INSTALL WIRE ASSY P/N 02333032-3 (J ASI 3) WIRES THROUGH THE SUPPLIED WIRE LOOM P/N P610079. ROUTE THE LOOM FROM THE LONGERON MOUNT P/N S2034-1 (SEE FIG. 11 AND REF. DWG 02333022) TO THE EXISTING GROUNDING POINT LOCATED BELOW AND FWD OF THE DOOR UNLOCK SWITCH. SECURE SUPPLIED WIRE LOOM TO WIRE MOUNT P/N 617-9452. SEE FIG. 12 AND REF DWG 02333024.</p> <p>36. TRIM WIRE (LA ASI 2) IF NECESSARY, AND INSTALL TERMINAL P/N MS25036-102. ATTACH GROUND WIRE (LA ASI 2) TO THE EXISTING GROUNDING POINT ON THE FWD CARGO DOOR JAMB. SEE FIG 13 AND REF DWG 02333024.</p> <p>37. CONTINUE ON WITH WIRE (LA ASI 1) PAST THE DOOR UNLOCK SWITCH, THROUGH THE EXISTING WIRE LOOM P/N P610079, TO THE CARGO DOOR LIGHT. TRIM WIRE IF NECESSARY AND INSTALL TERMINAL P/N MS25036-102. ATTACH WIRE (LA ASI 1) TO THE EXISTING TERMINAL ON THE CARGO LIGHT, REF DWG 02333024.</p> <p>38. RE-CONNECT BATTERY.</p> <p>39. TEST CARGO LIGHTS AND TIMER FOR PROPER OPERATION. SHUTOFF SHOULD OCCUR IN 30 MIN (±5 MIN). NOTE: ON AIRCRAFT WITH FACTORY CARGO LIGHT TIMERS INSTALLED, THE CARGO POD LIGHT SHUTOFF WILL BE COINCIDENT WITH THE FACTORY TIMER.</p> <p>40. REFER TO DWG 02333014 AND COMPLETE RE-INSTALLATION NOTES 11 THROUGH 15.</p> <p>41. CALCULATE AND RECORD NEW W&amp;B.</p> <p>42. INSERT SUPPLEMENT INTO AFM.</p> <p>43. RECORD INSTALLATION OF STC IN AIRCRAFT LOGS.</p>					
25	6	NAS1149F 0332P		WASHER	
24	1	AN525-10R18		SCREW	
23	1	NAS43HT3-32		SPACER	
22	1	MS9600-54		BRACKET	
21	2	AN3-4		BOLT	
20	3	S3923S3 (CESSNA P/N)		DOME NUT, CAD-PLATED STEEL, #10-32 THREAD	
19	2	NAS1149D0732		WASHER	
18	17	MS35333-39		INTERNAL STAR WASHER	
17	9	MS21919WDG4		ADEL CLAMP	
16	1	AN924-4D		BULKHEAD NUT	
15	1	AN837-4D		45 DEG FITTING	
14	1	6900329-000		TIMER, 30 MINUTE (T ASI 1)	
13	3	6900311-000		LIGHT (L ASI 1,2,3)	
12	2	6100295-000		CONNECTOR (P ASI 2,3)	
11	1	02333032-7		ASSY (WIRING HARNESS (J ASI 7))	
10	1	02333032-5		ASSY (WIRING HARNESS (J ASI 5))	
9	1	02333032-3		ASSY (WIRING HARNESS (J ASI 3))	
8	1	02333041-7		ALUMINUM TUBE	
7	1	02333041-5		ALUMINUM TUBE	
6	1	02333031-3		TUBE ASSY	
5	1	02333031-1		TUBE ASSY	
4	1	02333030-4		CTR MOUNT	
3	1	02333030-3		FWD MOUNT, FWD CARGO POD BAY	
2	3	02333030-2		AFT MOUNT	
1	2	02333030-1		FWD MOUNT	
		02333024-1		WIRE ASSEMBLY INSTALLATION, CESSNA 208	
		02333010-1		CARGO POD LIGHTING INSTALLATION, CESSNA 208	
ITEM	QTY	PART NUMBER	DESCRIPTION		
PARTS LIST					
DRAWN		AIRCRAFT STRUCTURES INTERNATIONAL CORP. (ASIC)			
M. BENSON		RR5, BOX 41B, ENID WOODRING REGIONAL AIRPORT			
CHECKED		ENID, OK 73701			
R. MITTEN		TITLE			
QA		CARGO POD LIGHTING INSTALLATION			
T. MARKES		CESSNA 208			
TOLERANCES (UNLESS SPECIFIED)		SIZE	DWG NO	REV	
		C	02333010	1	
HOLE DIAMETER: +.010/-0.000		SCALE	NONE	SHEET 1 OF 4	

APPENDIX A  
 Figure A-1

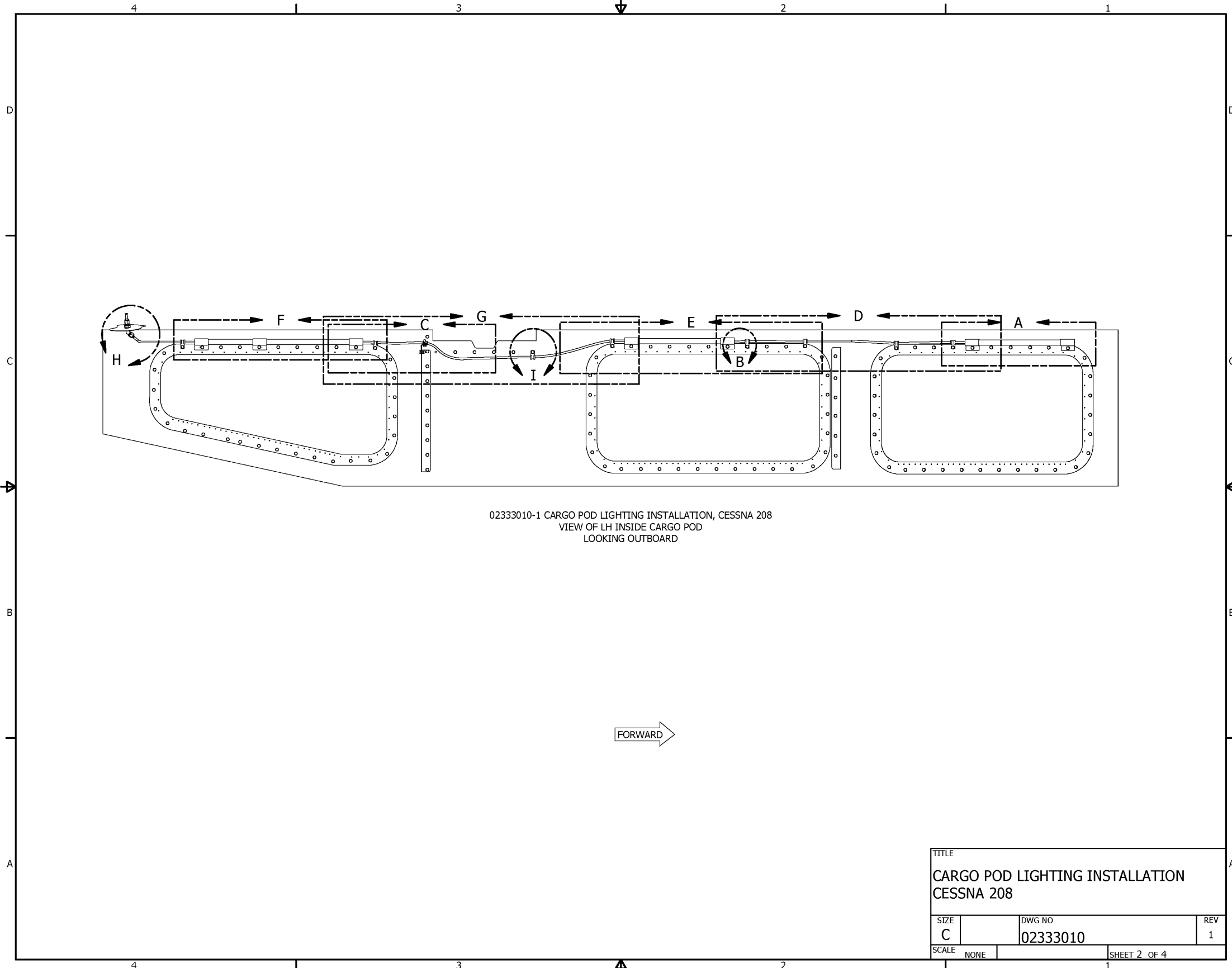


Figure A-2



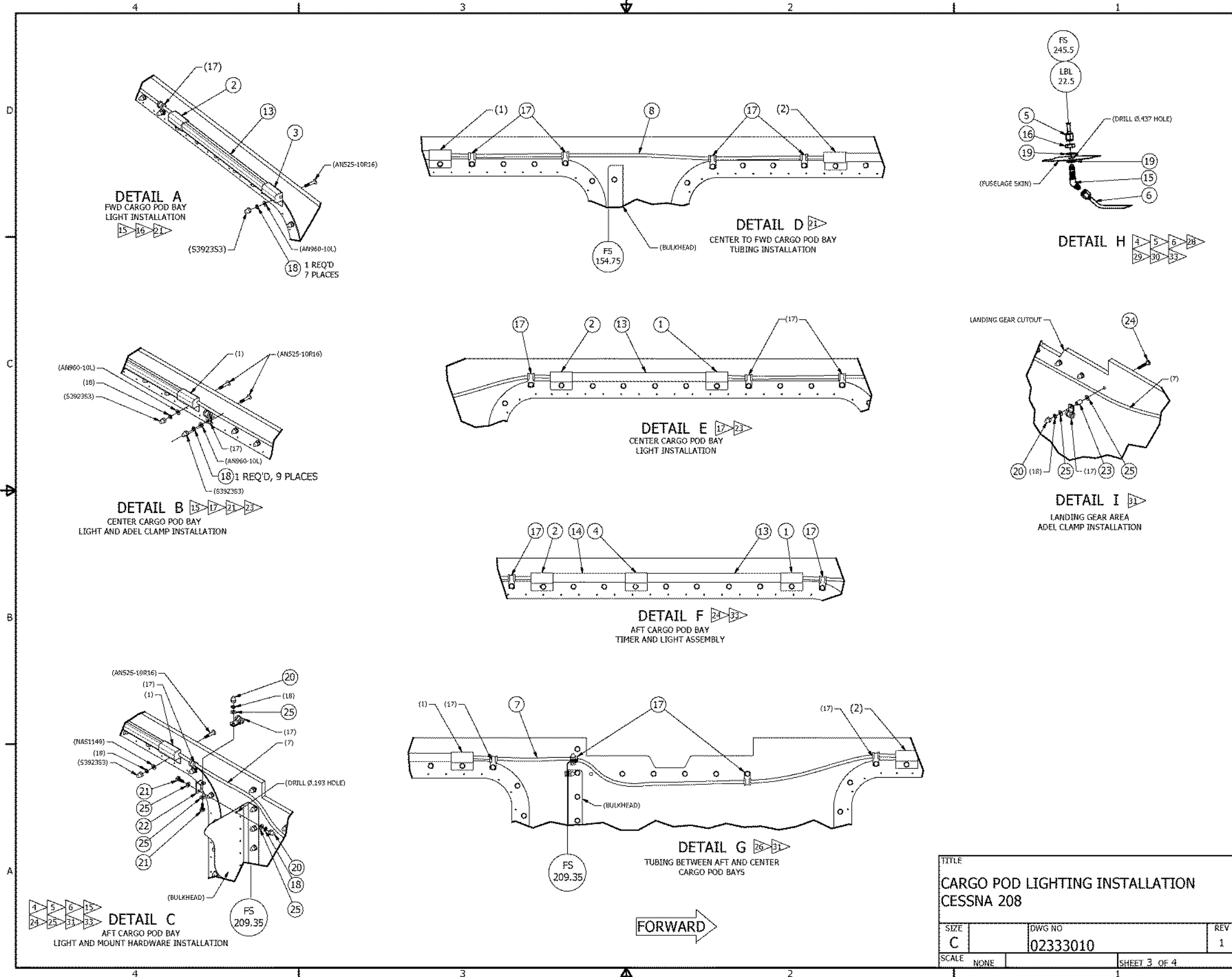


Figure A-3



TITLE		
CARGO POD LIGHTING INSTALLATION CESSNA 208		
SIZE	DWG NO	REV
C	02333010	1
SCALE	NONE	SHEET 4 OF 4

Figure A-4  
 Page 13 of 17



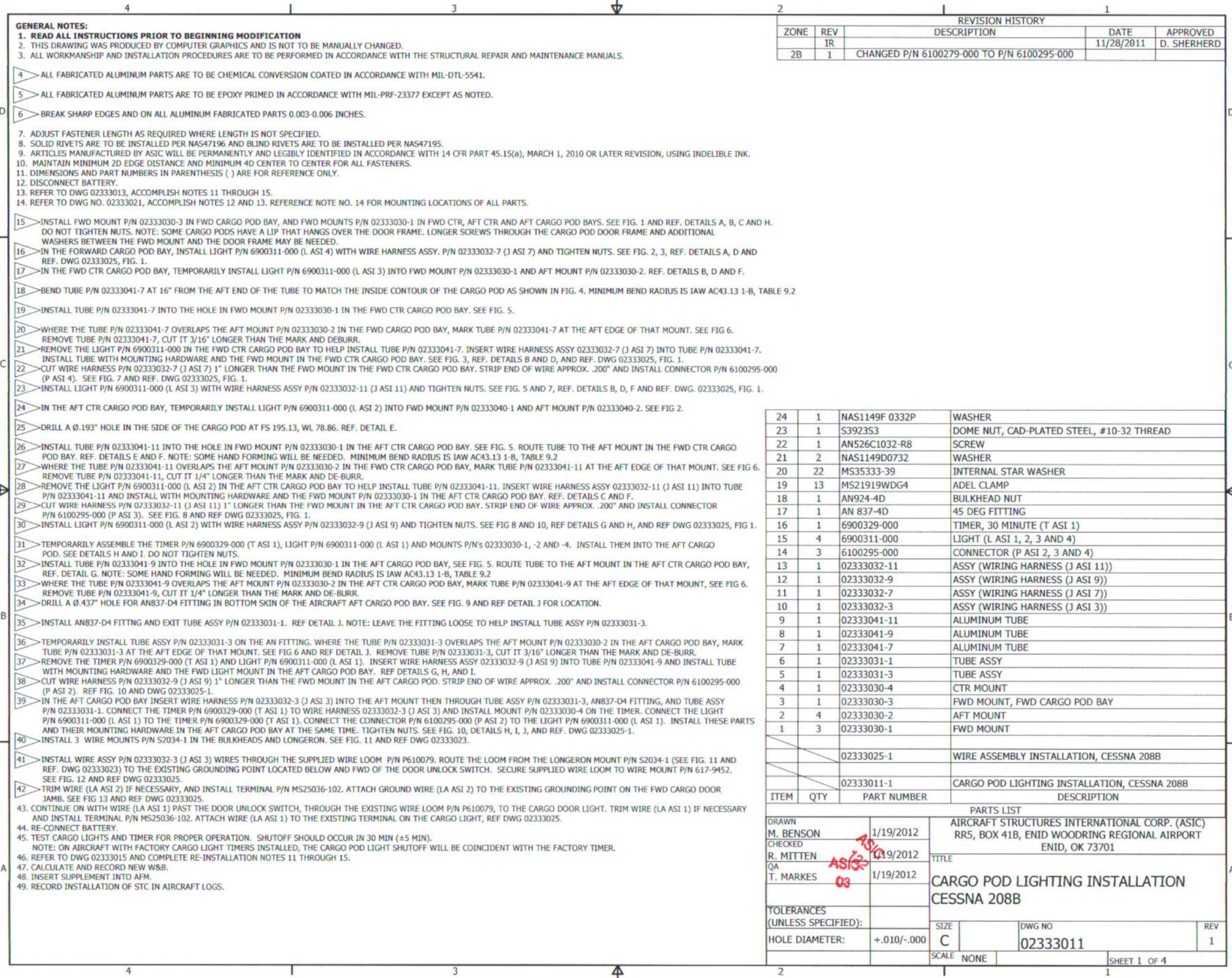


Figure A-5



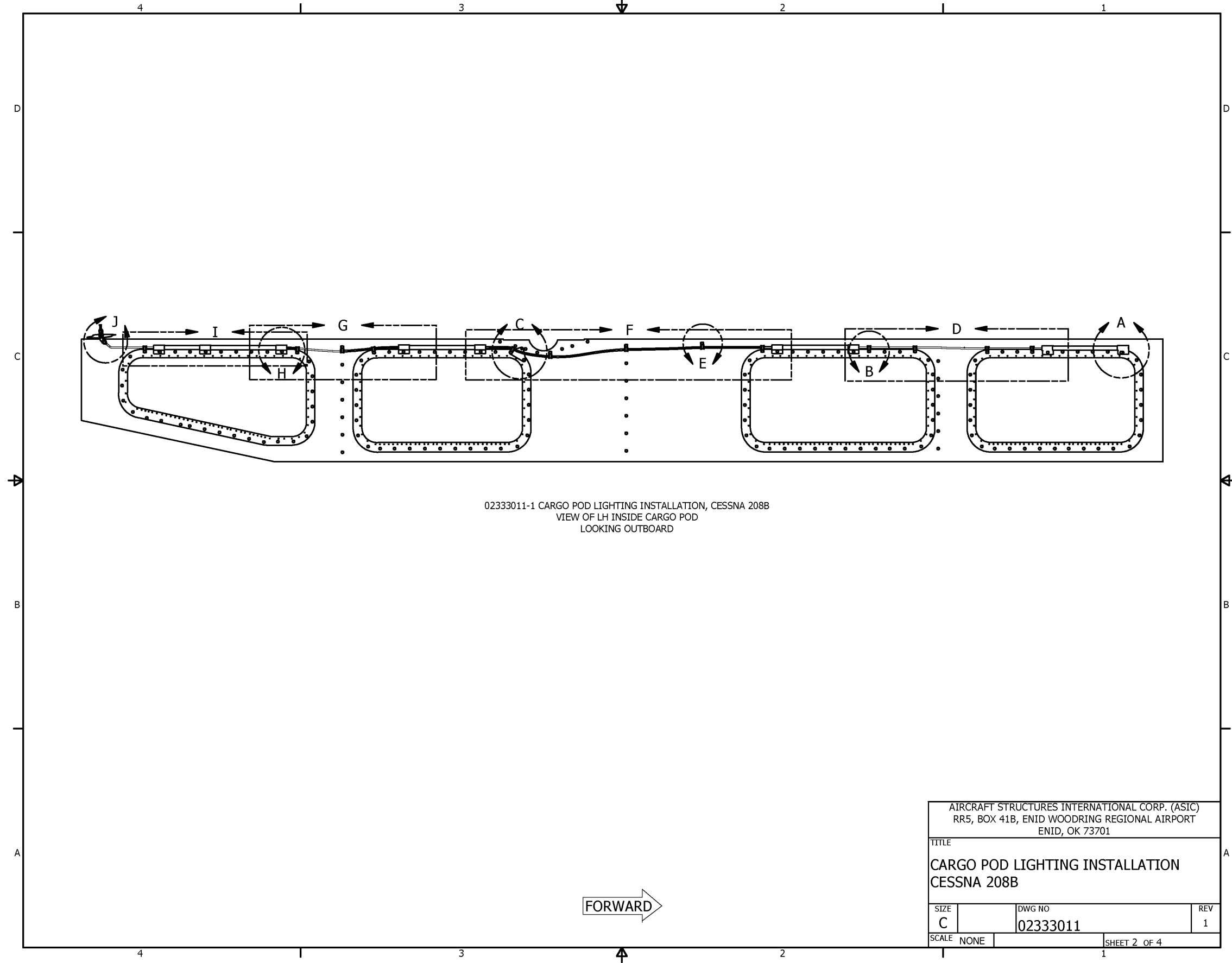


Figure A-6

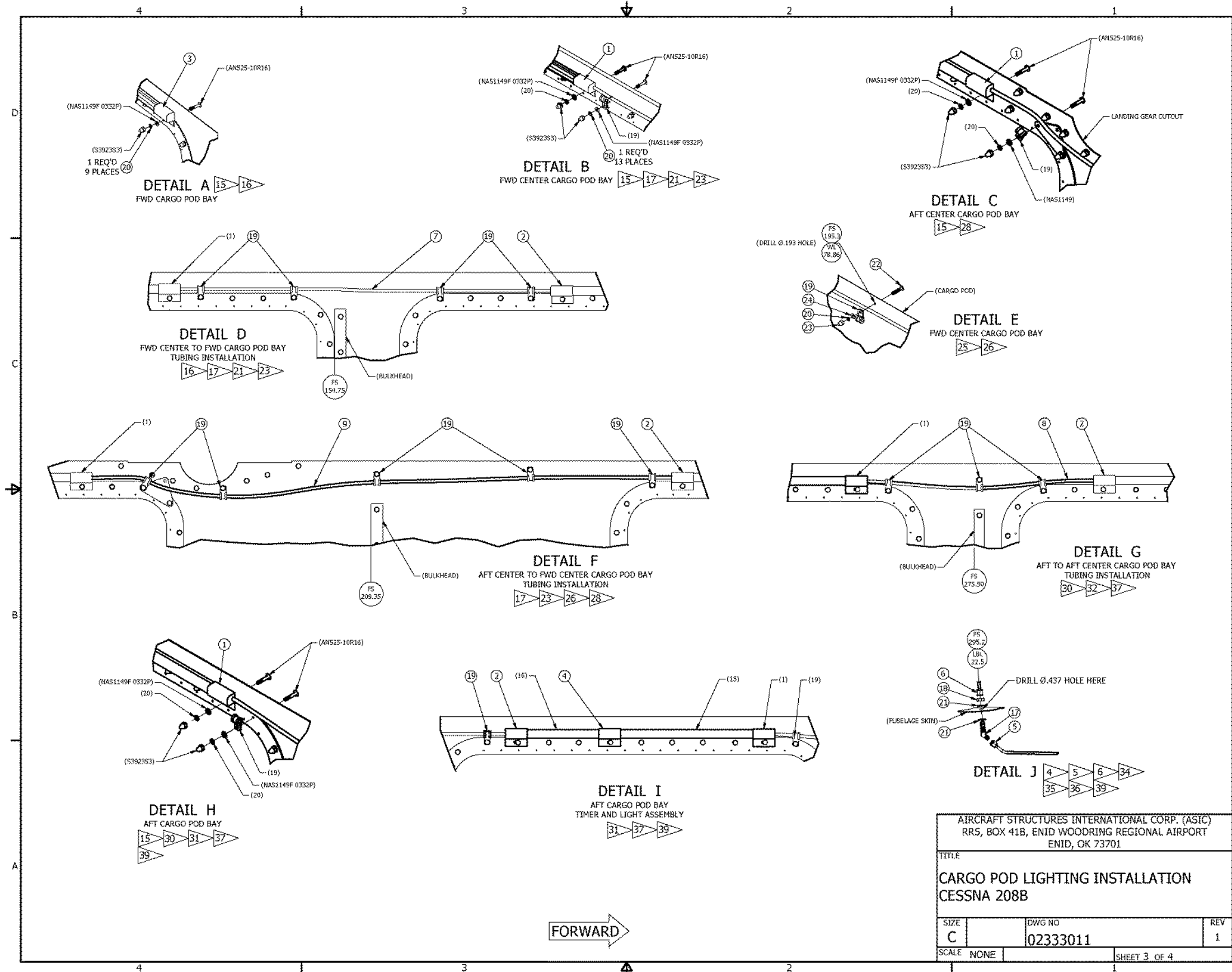
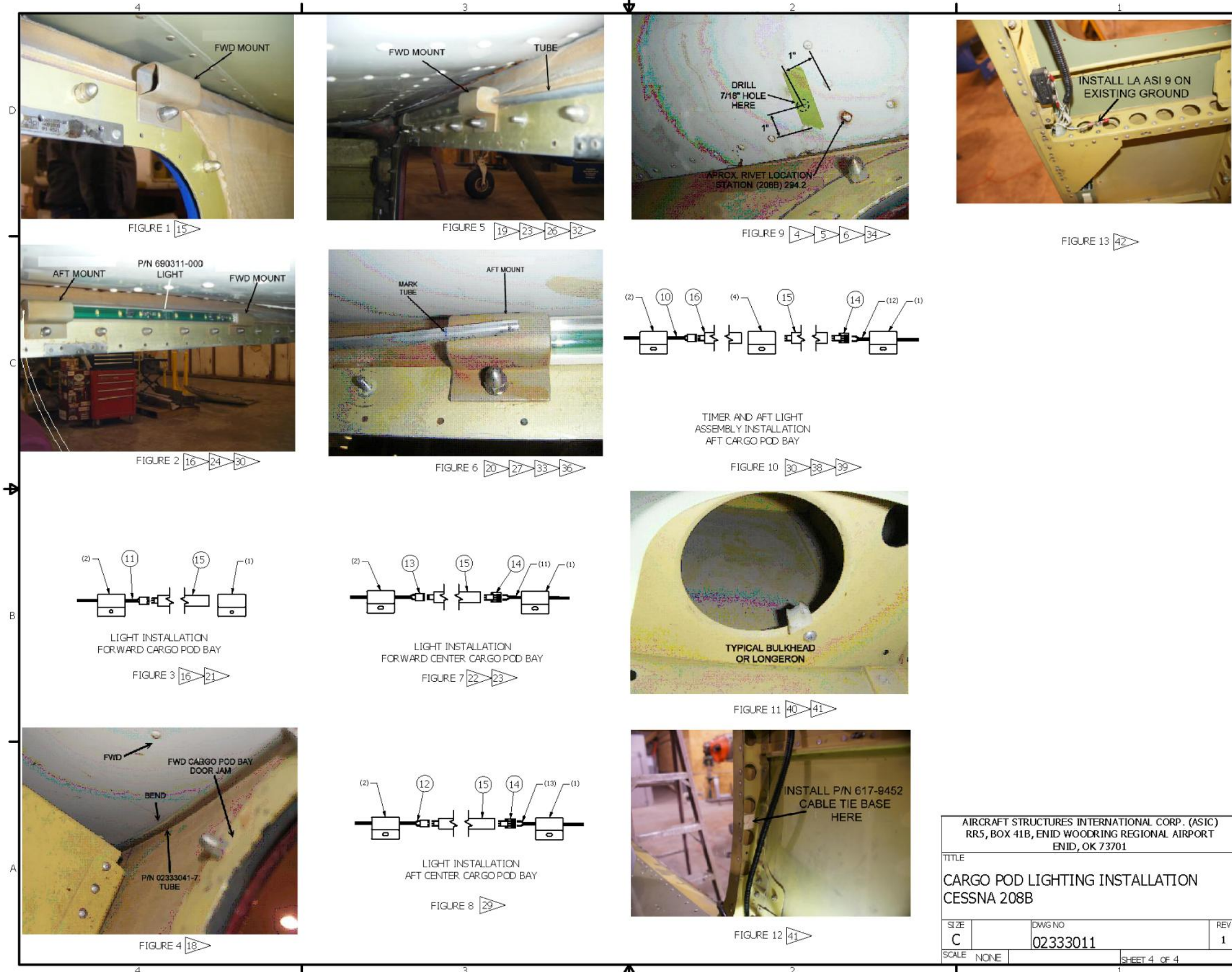


Figure A-7



AIRCRAFT STRUCTURES INTERNATIONAL CORP. (ASIC) RR5, BOX 41B, ENID WOODRING REGIONAL AIRPORT ENID, OK 73701		
TITLE CARGO POD LIGHTING INSTALLATION CESSNA 208B		
SIZE C	DWG NO 02333011	REV 1
SCALE NONE	SHEET 4 OF 4	

Figure A-8  
 Page 17 of 17